

TECHNICAL REPORT ARLCB-TR-80002

STRESS SINGULARITY AT THE VERTEX OF A FLAT WEDGE-SHAPED CRACK
BY A VARIATIONAL METHOD

M. A. Hussain
B. Noble
S. L. Pu

January 1980



US ARMY ARMAMENT RESEARCH AND DEVELOPMENT COMMAND
LARGE CALIBER WEAPON SYSTEMS LABORATORY
BENET WEAPONS LABORATORY
WATERVLIET, N. Y. 12189

AMCMS No. 611102H600211

DA Project No. 1L1161102H60

PRON No. 1A924324GGGG

APPROVED FOR PUBLIC RELEASE; DISTRIBUTION UNLIMITED

DTIC QUALITY INSPECTED 1

DISCLAIMER

The findings in this report are not to be construed as an official Department of the Army position unless so designated by other authorized documents.

The use of trade name(s) and/or manufacturer(s) does not constitute an official indorsement or approval.

DISPOSITION

Destroy this report when it is no longer needed. Do not return it to the originator.

REPORT DOCUMENTATION PAGE		READ INSTRUCTIONS BEFORE COMPLETING FORM
1. REPORT NUMBER ARLCB-TR-80002	2. GOVT ACCESSION NO.	3. RECIPIENT'S CATALOG NUMBER
4. TITLE (and Subtitle) STRESS SINGULARITY AT THE VERTEX OF A FLAT WEDGE-SHAPED CRACK BY VARIATIONAL METHOD		5. TYPE OF REPORT & PERIOD COVERED
7. AUTHOR(s) M. A. Hussain B. Noble S. L. Pu		6. PERFORMING ORG. REPORT NUMBER
9. PERFORMING ORGANIZATION NAME AND ADDRESS Benet Weapons Laboratory Watervliet Arsenal, Watervliet, NY 12189 DRDAR-LCB-TL		8. CONTRACT OR GRANT NUMBER(s)
11. CONTROLLING OFFICE NAME AND ADDRESS US Army Armament Research and Development Command Large Caliber Weapon Systems Laboratory Dover, New Jersey 07801		10. PROGRAM ELEMENT, PROJECT, TASK AREA & WORK UNIT NUMBERS AMCMS No. 611102H600211 DA Project No. 1L1161102H60 PRON No. 1A924324GGGG
14. MONITORING AGENCY NAME & ADDRESS (if different from Controlling Office)		12. REPORT DATE January, 1980
		13. NUMBER OF PAGES 30
		15. SECURITY CLASS. (of this report) UNCLASSIFIED
		15a. DECLASSIFICATION/DOWNGRADING SCHEDULE
16. DISTRIBUTION STATEMENT (of this Report) Approved for public release; distribution unlimited.		
17. DISTRIBUTION STATEMENT (of the abstract entered in Block 20, if different from Report)		
18. SUPPLEMENTARY NOTES Presented at 1979 ASME-CSME Applied Mech/Bioengineering Conf/Fluid Eng Conf, Niagara Hilton Hotel, Niagara Falls, NY, 18-20 June 1979. Presented at 2nd MACSYMA Users' Conf, Washington, DC, 20-22 June 79. To be published in an open literature journal.		
19. KEY WORDS (Continue on reverse side if necessary and identify by block number) Variational Method Spherical Harmonics Flat-Wedge-Shaped Crack Mixed Boundary Value Problems Stress Intensity Factors Fracture Mechanics ~ <i>crack propagation</i> Papkovitch Stress Functions		
20. ABSTRACT (Continue on reverse side if necessary and identify by block number) Three dimensional elasticity problems are generally complex. In this paper we present the analysis for the stress singularity at the apex of a three dimensional, flat, wedge-shaped crack under general loadings. The problem is reduced to a set of coupled dual integral equations. Because of the complexity they are not amenable to a closed form solution. A variational method is developed to handle such problems. The physical interpretation of the results is also presented.		

SECURITY CLASSIFICATION OF THIS PAGE(When Data Entered)

SECURITY CLASSIFICATION OF THIS PAGE(When Data Entered)

TABLE OF CONTENTS

	<u>Page</u>
INTRODUCTION	1
VARIATIONAL THEOREM	2
PAPKOVITCH STRESS FUNCTIONS	4
THREE MODES OF CRACKS AND COUPLED INTEGRAL EQUATIONS	6
MODE I	7
MODE II	8
MODE III	8
APPLICATION OF VARIATIONAL PRINCIPLE	11
CONCLUSIONS	14
REFERENCES	15
APPENDIX 1	16
APPENDIX 2	19

TABLES

I. BOUNDARY CONDITIONS ON $\theta = \pi/2$	7
II. VALUES OF μ FOR $\nu = 0.25$	13

ILLUSTRATIONS

1. A flat wedge-shaped crack under three different modes and its two-dimensional counterparts.	23
2. A spherical coordinate surrounding the apex of a thin angular sector crack.	24
3. The eigenvalue μ as a function of α/π for $\nu = 0.25$.	
(a) Variational results indicated by x and • for mode II.	25
(b) Variational results indicated by x and • for mode III.	26

INTRODUCTION

The theory of fracture mechanics has been a very successful tool in engineering application in recent years. This is mainly due to the use of a single characteristic parameter namely the stress intensity factor, that is the coefficient of the stress singularity at the tip of a crack in the linear theory of elasticity. In most of the two dimensional cracks, in homogeneous media, the singularity is of the order one half. For the three dimensional cracks, however, the singularity depends upon the geometric configurations.

In this report we study the singularities at the apex of a thin wedge-shaped crack shown in Figure 1 under three loading conditions. Using the near field approach, the problem is reduced to an eigenvalue problem for coupled dual integral equations. The results indicate that cracks tend to straighten out at the apex.

In the following sections we first prove the variational theorem by which the eigenvalue of coupled integral equations is to be obtained. This eigenvalue problem is not the linear one commonly encountered in mathematical physics. Next we present Papkovitch stress function approach to three dimensional theory of elasticity. Then coupled dual series relations are obtained by using mixed boundary conditions. These series are transformed to coupled singular integral equations. Finally the variational method is applied to the coupled integral equations to obtain eigenvalues.

The variational method completely avoids the solution of complex singular integral equations. In this study Macsyma was found to be an indispensable tool at all levels of analysis.

VARIATIONAL THEOREM

Consider the following pair of homogeneous coupled integral equations with Fredholm kernels.

$$\int_0^{\alpha} K_{11}(\phi, \psi; \mu) f(\psi) d\psi + \int_0^{\alpha} K_{12}(\phi, \psi; \mu) g(\psi) d\psi = 0 \quad (1)$$

$$\int_0^{\alpha} K_{21}(\phi, \psi; \mu) f(\psi) d\psi + \int_0^{\alpha} K_{22}(\phi, \psi; \mu) g(\psi) d\psi = 0 \quad (2)$$

where eigenvectors f and g and the eigenvalue μ are unknown and $K_{11}(\phi, \psi; \mu)$ etc. involve μ in a linear or nonlinear fashion. Construct the following characteristic equation for the determination of μ^* with appropriate trial functions $f^*(\psi)$ and $g^*(\psi)$.

$$\begin{aligned} & \left(\int_0^{\alpha} f^*(x) \int_0^{\alpha} K_{11}(x, y) f^*(y) dy dx \right) \left(\int_0^{\alpha} g^*(\phi) \int_0^{\alpha} K_{22}(\phi, \psi) g^*(\psi) d\psi d\phi \right) \\ & - \left(\int_0^{\alpha} f^*(x) \int_0^{\alpha} K_{12}(x, y) g^*(y) dy dx \right) \left(\int_0^{\alpha} g^*(\phi) \int_0^{\alpha} K_{21}(\phi, \psi) f^*(\psi) d\psi d\phi \right) = 0 \end{aligned} \quad (3)$$

If f^* and g^* vary around the exact solutions f and g as

$$f^*(\phi) = f(\phi) + \delta \xi(\phi) \quad , \quad g^*(\phi) = g(\phi) + \delta \eta(\phi) \quad (4)$$

then $(\mu^* - \mu)$ is stationary around δ as δ approaches zero. That is

$$\mu^* = \mu + O(\delta^2)$$

provided

$$\begin{aligned}
K_{11}(\phi, \psi) &= K_{11}(\psi, \phi) \\
K_{22}(\phi, \psi) &= K_{22}(\psi, \phi) \\
K_{12}(\phi, \psi) &= K_{21}(\psi, \phi)
\end{aligned} \tag{5}$$

Proof: Symbolically we write equation (3) as

$$(f^* K_{11} f^*)(g^* K_{22} g^*) - (f^* K_{12} g^*)(g^* K_{21} f^*) = 0 \tag{6}$$

where

$$f^* = f + \delta \xi, \quad g^* = g + \delta \eta. \tag{7}$$

Expanding the kernels around μ , we have

$$K_{11}(\mu) = K_{11}(\mu^*) + \Delta K'_{11}(\mu^*) + O(\Delta^2), \quad \text{etc.} \tag{8}$$

where

$$\Delta = \mu - \mu^* \tag{9}$$

Substituting from (7) and (8) into (6) and using (5) we obtain

$$\begin{aligned}
&[(fK_{11}f)(gK_{22}g) - (fK_{12}g)(fK_{12}g)] \\
&+ 2\delta[(\xi K_{11}f)(gK_{22}g) - (\xi K_{12}g)(fK_{12}g) + (\eta K_{22}g)(fK_{11}f) - (fK_{12}\eta)(fK_{12}g)] \\
&+ L_1 \Delta + L_2 \delta^2 + \text{higher order terms} = 0
\end{aligned} \tag{10}$$

where

$$L_1 = (fK_{11}f)(gK'_{22}g) + (fK'_{11}f)(gK_{22}g) - 2(fK_{12}g)(fK'_{12}g) \tag{10a}$$

$$\begin{aligned}
L_2 &= (\xi K_{11}\xi)(gK_{22}g) - (\xi K_{12}g)^2 + 4(fK_{11}\xi)(gK_{22}\eta) \\
&+ (fK_{11}f)(\eta K_{22}\eta) - (fK_{12}\eta)^2 - 4(fK_{12}\eta)(\xi K_{12}g).
\end{aligned} \tag{10b}$$

Using equations (1) and (2) it is seen that the first two terms in (10) vanish.

Hence

$$\Delta = \delta^2(L_2/L_1) + O(\delta^3). \tag{11}$$

The above proves that Δ is of the second order in δ . This completes the proof of the theorem. Because the eigenvalues involved in kernels can be in a nonlinear fashion we cannot prove its bounds as can be done for the linear case in Reference 1.

PAPKOVITCH STRESS FUNCTIONS

In the absence of body force the equation of equilibrium of a homogeneous isotropic elastic solid is given by

$$\nabla^2 \bar{u} + \frac{1}{1-2\nu} \nabla \nabla \cdot \bar{u} = 0 \quad (12)$$

Using Helmholtz decomposition theorem the solution of (12) can be written as^{2,3}

$$2G\bar{u} = \nabla B_0 + \nabla(\bar{r} \cdot \bar{B}) - 4(1-\nu)\bar{B} \quad (13)$$

where G and ν are shear modulus and Poisson's ratio, B_0 and \bar{B} are known as Papkovitch stress functions satisfying

$$\nabla^2 B_0 = 0, \quad \nabla^2 \bar{B} = 0, \quad \bar{B} = i\psi + j\omega + k\lambda \quad (14)$$

For computational purposes it is convenient to write the solution as a superposition of the following basic solutions.

$$\begin{aligned} \text{1st Basic solution:} \quad & 2G\bar{u} = \nabla B_0 \\ \text{2nd Basic solution:} \quad & 2G\bar{u} = \nabla(x\psi) - 4(1-\nu)\psi\bar{i} \\ \text{3rd Basic solution:} \quad & 2G\bar{u} = \nabla(y\omega) - 4(1-\nu)\omega\bar{j} \\ \text{4th Basic solution:} \quad & 2G\bar{u} = \nabla(z\lambda) - 4(1-\nu)\lambda\bar{k} \end{aligned} \quad (15)$$

¹Barlett, C.C. and Noble, B., "A Variational Method for the Solution of Eigenvalue Problems Involving Mixed Boundary Conditions," Applied Science Research, Section B, Vol. 9, 1962.

²Green, A.E. and Zerna, W., THEORETICAL ELASTICITY, 2nd Edition, Oxford, 1968.

³Lure', A.T., THREE DIMENSIONAL PROBLEMS OF THE THEORY OF ELASTICITY, Inter-Science Publishers, 1964.

These basic solutions were transformed to the following spherical coordinate system

$$x = r \sin \theta \cos \phi, \quad y = r \sin \theta \sin \phi, \quad z = r \cos \theta \quad (16)$$

with the origin at the apex of the crack as shown in Figure 2. This enables us to obtain near field solutions and to study the stress singularity at the apex. The complete results of components of displacement and stress for basic solutions are given in Appendix 1. Since we are interested in the power singularity at the apex, so we choose Papkovitch potentials in the form

$$H(r, \theta, \phi) = r^\mu H_1(\theta, \phi) \quad (17)$$

where μ is the eigenvalue to be determined. As can be seen from Appendix 1, stress components will be of the form

$$\sigma = O(r^{\mu-1}) \quad (18)$$

which will be singular when $\mu < 1$. For the displacements to be finite we seek positive eigenvalues between 0 and 1.

The near field geometry surrounding the apex permits us to write (17) as the separation of variables solution

$$H(r, \theta, \phi) = \sum_m r^\mu P_\mu^m(\cos \theta) \begin{matrix} \cos m\phi \\ \text{or} \\ \sin m\phi \end{matrix} \quad (19)$$

where $P_\mu^m(x)$ is the associated Legendre function of the first kind with degree μ and order m . When (19) is substituted into the four basic solutions given in Appendix 1 we see that the forms of some solutions thus obtained are not convenient to work with. The final solutions used in this analysis are designated as solutions A, B, C, and D which are given in Appendix 2. Solution A was obtained by replacing μ by $\mu + 1$ after the substitution from (19)

into the first basic solution. Solution B is obtained by adding the second and the third basic solutions with proper trigonometric functions in (19), replacing m by $m + 1$ and then using Legendre recursion formulae. A similar process was used to obtain solution C. Solution D simply comes from the fourth basic solution with the use of (19).

Solutions A, B, C, and D must not be linearly independent. This is due to the fact that the condition that the vector point function is solenoidal has not been used explicitly in Papkovitch stress functions approach. We found these solutions are indeed not independent. A relation among them can be written symbolically in the form

$$(\mu+m+1)[C] = [B] + 2(\mu+m+1)[D] - 2(\mu-3+4\nu)[A] \quad . \quad (20)$$

Hereafter solution C is replaced by solutions A, B, D using (20).

THREE MODES OF CRACKS AND COUPLED INTEGRAL EQUATIONS

For a crack shown in Figure 2, the leading edges of the crack are $\phi = \pm \alpha$ and the crack is in the x - y plane ($\theta = \pi/2$). Let D^- and D^+ be the cracked and uncracked region of the plane $\theta = \pi/2$. Within the cracked region, the displacement is discontinuous. If the discontinuity is in the z -direction ($u_\theta^+ - u_\theta^- = \text{finite}$), the crack is under mode I; if the discontinuity is in the x -direction ($u_x^+ - u_x^- = \text{finite}$), the crack is defined to be under mode II; and if $u_y^+ - u_y^- = \text{finite}$, the crack is defined to be under mode III. Boundary conditions for various modes are tabulated below.

TABLE I. BOUNDARY CONDITIONS ON $\theta = \pi/2$

	Non Mixed Conditions (in $D^- + D^+$)	Mixed Conditions in D^- in D^+	
Mode I	$\tau_{\theta r} = \tau_{\theta \phi} = 0$	$\sigma_{\theta} = 0$	$u_{\theta} = 0$ (21)
Mode II	$\sigma_{\theta} = 0$	$\tau_{\theta r} = \tau_{\theta \phi} = 0$	$u_r = u_{\phi} = 0$ (22)
Mode III	$\sigma_{\theta} = 0$	$\tau_{\theta r} = \tau_{\theta \phi} = 0$	$u_r = u_{\phi} = 0$ (23)

For mode I, u_{θ} is even in ϕ . This leads to the use of the trigonometric functions at the top of (19). The boundary conditions of (22) and (23) are identical, but the symmetric properties are different for mode II and mode III. In the former case, u_r is even and u_{ϕ} is odd in ϕ while in the latter, the reverse is true. Hence the proper set of quantities should be selected in (19), for each case.

MODE I. Using equation (19) and the non-mixed conditions of (21), we have

$$B_m = 0, \quad A_m = (m+\mu+1)^{-1} (1-2\nu) D_m. \quad (24)$$

The mixed boundary conditions of (21) and using (24) and (19), yield

$$\begin{aligned} \sum b_m \cos m\phi &= 0 & 0 \leq \phi < \alpha \\ \sum q_m b_m \cos m\phi &= 0 & \alpha < \phi \leq \pi \end{aligned} \quad (25)$$

where \sum denotes the summation with respect to m for $m = 0, 1, 2, \dots, \infty$, and

$$b_m = (-m+\mu+1) D_m P_{\mu+1}^m, \quad q_m = (-m+\mu+1)^{-1} P_{\mu}^m / P_{\mu+1}^m, \quad P_{\mu}^m = P_{\mu}^m(0) \quad (26)$$

MODE II. For the homogeneous condition of σ_θ in (22), using (19), the coefficients A, B, and D are related in the form

$$2(1-\nu)D_m = (m+\mu+1)A_m + [\mu^2 + 2\nu(\mu+1) - 2(1-\nu)m - 1]B_m$$

This relation and the mixed conditions of (22) yield the following coupled dual series:

$$\begin{aligned} \sum E_m \cos m\phi &= 0 & 0 \leq \phi < \alpha \\ \sum (R_m E_m + S_m F_m) \cos m\phi &= 0 & \alpha < \phi \leq \pi \end{aligned} \quad (27)$$

$$\begin{aligned} \sum' F_m \sin m\phi &= 0 & 0 \leq \phi < \alpha \\ \sum' (U_m F_m + T_m E_m) \sin m\phi &= 0 & \alpha < \phi \leq \pi \end{aligned} \quad (28)$$

where \sum' denotes the summation with respect to m for $m = 1, 2, \dots, \infty$, and

$$E_m = -(m+\mu+1)\{\mu A_m + [4m(1-\nu)^2 + \mu(\mu+4\nu-m-3)]E_m\}P_{\mu}^m \quad (29)$$

$$F_m = (m+\mu+1)\{m A_m + [4\mu(1-\nu)^2 + m(\mu+4\nu-m-3)]B_m\}P_{\mu}^m \quad (30)$$

$$\begin{aligned} R_m &= [m^2 - \mu(\mu+1)(1-\nu)]V_m, & S_m &= m(1-\nu-\nu\mu)V_m \\ U_m &= [(1-\nu)m^2 - \mu(\mu+1)]V_m, & T_m &= m(1+\nu\mu)V_m \end{aligned} \quad (31)$$

in (31) V_m stands for $P_{\mu+1}^m/P_{\mu}^m/[(m+\mu+1)(m^2-\mu^2)]$.

MODE III. Similar to the preceding case, we have

$$\begin{aligned} \sum' E_m \sin m\phi &= 0 & 0 \leq \phi < \alpha \\ \sum' (R_m E_m + S_m F_m) \sin m\phi &= 0 & \alpha < \phi \leq \pi \end{aligned} \quad (32)$$

$$\begin{aligned} \sum F_m \cos m\phi &= 0 & 0 \leq \phi < \alpha \\ \sum (U_m F_m + T_m E_m) \cos m\phi &= 0 & \alpha < \phi \leq \pi \end{aligned} \quad (33)$$

It was found that the dual series (25) for mode I is identical to that of a potential problem studied in References 4, 5, and 6 and will not be discussed here.

For convenience we make the following change of variables

$$\begin{aligned}\phi &= \pi - \omega, \quad \alpha' = \pi - \alpha, \quad (-1)^m E_m = E_m', \quad (-1)^m F_m = F_m' \\ R_m &= R_m' (1 - \nu - \nu\mu), \quad S_m = S_m' (1 - \nu - \nu\mu) \\ U_m &= U_m' (1 + \nu\mu) (-1), \quad T_m = T_m' (1 + \nu\mu) (-1).\end{aligned}\quad (34)$$

The dual series of (27), (28) now become

$$\sum E_m' \cos m\omega = 0 \quad \alpha' < \omega \leq \pi \quad (35)$$

$$\sum F_m' \sin m\omega = 0 \quad (36)$$

$$\sum (R_m' E_m' + S_m' F_m') \cos m\omega = 0 \quad 0 \leq \omega < \alpha' \quad (37)$$

$$\sum (T_m' E_m' + U_m' F_m') \sin m\omega = 0 \quad (38)$$

Let the right hand sides of (35) and (36) for the interval $0 < \omega < \alpha'$ be denoted by unknown functions $f(\omega)$ and $g(\omega)$, respectively. The Fourier inversion gives

$$\begin{aligned}E_0' &= \frac{1}{\pi} \int_0^{\alpha'} f(\psi) d\psi, & E_m' &= \frac{2}{\pi} \int_0^{\alpha'} f(\psi) \cos m\psi d\psi \\ F_m' &= \frac{2}{\pi} \int_0^{\alpha'} g(\psi) \sin m\psi d\psi\end{aligned}\quad (39)$$

⁴Noble, B., "The Potential and Charge Distribution Near the Tip of a Flat Angular Sector," EM-135, New York University, NY, 1959.

⁵Brown, S.N., and Stewartson, K., "Flow Near the Apex of a Plane Delta Wing," Journal of Institute of Mathematics and Its Applications, Vol. 5, p. 206, 1969.

⁶Morrison, J.A., and Lewis, J.A., "Charge Singularity at the Corner of a Flat Plate," SIAM, Journal of Applied Mathematics, Vol. 31, p. 233, 1976.

Substituting from (39) into (37), (38) and interchanging the order of summation and integration we have the following coupled integral equations for the determination of f and g ,

$$\int_0^{\alpha'} K_{11}(\omega, \psi; \mu) f(\psi) d\psi + \int_0^{\alpha'} K_{12}(\omega, \psi; \mu) g(\psi) d\psi = 0 \quad (40)$$

$$\int_0^{\alpha'} K_{21}(\omega, \psi; \mu) f(\psi) d\psi + \int_0^{\alpha'} K_{22}(\omega, \psi; \mu) g(\psi) d\psi = 0 \quad (41)$$

where

$$\begin{aligned} K_{11}(\omega, \psi; \mu) &= \frac{1}{2} R_0' + \sum R_m' \cos m\omega \cos m\psi \\ K_{12}(\omega, \psi; \mu) &= \sum S_m' \cos m\omega \sin m\psi \\ K_{21}(\omega, \psi; \mu) &= \sum T_m' \sin m\omega \cos m\psi \\ K_{22}(\omega, \psi; \mu) &= \sum U_m' \sin m\omega \sin m\psi \end{aligned} \quad (42)$$

It can be shown that⁷

$$\frac{p_{\mu+1}^m}{p_{\mu}^m} = \frac{-2}{m-\mu-1} \Gamma\left(\frac{m-\mu+1}{2}\right) \Gamma\left(\frac{m+\mu+2}{2}\right) / \Gamma\left(\frac{m-\mu}{2}\right) / \Gamma\left(\frac{m+\mu+1}{2}\right) \quad (43)$$

Using (43) and (34) the following asymptotic expansions can be established.

$$\begin{aligned} R_m' &= -(1-\nu-\mu\nu)^{-1} (1/m) + O(1/m^2) \quad , \quad S_m' = 1/m^2 + O(1/m^3) \\ U_m' &= [-(1-\nu)/(1+\nu\mu)] (1/m) + O(1/m^2) \quad , \quad T_m' = 1/m^2 + O(1/m^3) \end{aligned} \quad (44)$$

Substituting from (44) into (42) and summing the dominant part of the series we have,^{4,8}

⁴Noble, B., "The Potential and Charge Distribution Near the Tip of a Flat Angular Sector," EM-135, New York University, NY.

⁷Magnus, W. and Oberhettinger, F., SPECIAL FUNCTIONS OF MATHEMATICAL PHYSICS, Chelsea Publishing Co., 1949.

⁸Jolly, L.B.W., SUMMATION OF SERIES, Dover Publications, p. 126, 1961.

$$K_{11}(\omega, \psi) = \frac{1}{1-\nu-\nu\mu} \frac{1}{2} \log 2 |\cos \omega - \cos \psi| + \text{regular terms} \quad (45)$$

$$K_{22}(\omega, \psi) = \frac{-(1-\nu)}{2(1+\nu\mu)} \log \left| \sin\left(\frac{\omega+\psi}{2}\right) / \sin\left(\frac{\omega-\psi}{2}\right) \right| + \text{regular terms} \quad (46)$$

The previous expressions show the kernels have logarithmic singularities. Similar analysis can be carried out for equations (32) and (33) for mode III (interchanging R_m' and U_m' , S_m' and T_m').

APPLICATION OF VARIATIONAL PRINCIPLE

Equations (40) and (41) are identical to equations (1) and (2) and all the conditions of the theorem required for the kernels are satisfied. In this section we shall apply the theorem to obtain approximate eigenvalues by assuming approximate trial functions. Without causing ambiguity we shall drop the asterisks and assume the following trial functions,

$$\begin{aligned} f(t) &= (\beta_0 + \cos t) \cos(t/2) (\cos t - \cos \alpha')^{1/2} \\ g(t) &= [(1 - \cos \alpha') + 2 \cos t] \sin(t/2) / (\cos t - \cos \alpha')^{1/2} \end{aligned} \quad (47)$$

where

$$\beta_0 = -\cos^2(\alpha'/2) \frac{1 + (1-\nu-\nu\mu)R_0' + \log \sin^2(\alpha'/2)}{(1-\nu-\nu\mu)R_0' + \log \sin^2(\alpha'/2)} \quad (48)$$

The above trial functions are the first approximations to the integral equations (40) and (41) with kernels (42) replaced by their dominant parts given by (44). The method of obtaining such solutions by direct computation is illustrated in Reference 9.

⁹Noble, B., Hussain, M.A., and Pu, S.L., "Apex Singularities for Corner Cracks Under Opening, Sliding, and Tearing Modes," to be published in the Proceedings of the International Conference on Fracture Mechanics in Engineering Application, Bangalore, India, 1979.

Substituting from (47) into the characteristic equation (3), changing the order of summation and integration and using the integral representation of Legendre functions,⁷ we obtain

$$I_{11}I_{22} - I_{12}^2 = 0 \quad (49)$$

where

$$\begin{aligned} I_{11} &= 2R_o' (1+2\beta_o + P_1)^2 + \sum' R_m' [P_{m+1} + (1+2\beta_o)(P_m + P_{m+1}) + P_{m-2}]^2 \\ I_{22} &= 4 \sum' U_m' [P_{m+1} - P_1(P_m - P_{m-1}) - P_{m-2}]^2 \\ I_{12} &= -2 \sum' S_m' [P_{m+1} + (1+2\beta_o)(P_m + P_{m-1}) + P_{m-2}] [P_{m+1} - P_1(P_m - P_{m-1}) - P_{m-2}] \\ P_m &= P_m(\cos \alpha') \end{aligned} \quad (50)$$

For different values of ν and α' (the complementary angle to half of the vertex angle of the wedge-shaped crack) we get approximate values of μ from equation (49) for both modes II and III.

The form of (49) is very well suited for Macsyma evaluation and using eight terms for summations in (50), the results for μ are marked by x in Figure 3 where the solid lines are results obtained by another method.⁹ The results by both methods are in good agreement.

A further refinement of results can be obtained by selecting

$$\begin{aligned} f(t) &= (A + B \cos t) \frac{\cos(t/2)}{(\cos t - \cos \alpha')^{1/2}} \\ g(t) &= (C + D \cos t) \frac{\sin(t/2)}{(\cos t - \cos \alpha')^{1/2}} \end{aligned} \quad (51)$$

⁷Magnus, W. and Oberhettinger, F., SPECIAL FUNCTIONS OF MATHEMATICAL PHYSICS, Chelsea Publishing Co., 1949.

and formally extending the variational technique to a characteristic equation obtained from the vanishing of the determinant of a four by four system (A, B, C, D in (51) must not all vanish). The results, using Macsyma and summing to a maximum of eight terms in (50), are compared in the following table. They also are shown as \cdot in Figure 3.

TABLE II. VALUES OF μ FOR $\nu = 0.25$

Mode	Half Vertex Angle α	Variational Method		Direct Method Ref. 9
		Using (47)	Using (51)	
II	0.1π	0.9333	0.9616	0.9582
	0.3π	0.6489	0.6961	0.6953
	0.5π	0.4752	0.5107	0.5017
	0.7π	0.3585	0.3749	0.3654
	0.9π	0.2213	0.2469	0.2137
III	0.1π	0.9739		
	0.3π	0.7642	0.8253	0.8270
	0.5π	0.4881	0.5192	0.5027
	0.6π	0.3829	0.4034	0.3914
	0.7π	0.3021		0.3015
	0.8π	0.2372		0.2335

⁹Noble, B., Hussain, M.A., and Pu, S.L., "Apex Singularities for Corner Cracks Under Opening, Sliding, and Tearing Modes," to be published in the Proceedings of the International Conference on Fracture Mechanics in Engineering Application, Bangalore, India, 1979.

Even the rigorous proof of the extension of the variational technique from the two by two system using trial functions (47) to the four by four system using trial functions (51) is still to be done, the results thus obtained are in good agreement with results achieved by using trial functions (47) or other methods.⁹

CONCLUSIONS

For modes II and III, the results show that the stress singularities are dominated by the vertex angle as well as the elastic constant ν of the material. The results further indicate that when the apex angle is greater than 180° , the stress singularity is stronger than one half enhancing the tendency of crack front to straighten out. Similarly, when the vertex angle is less than 180° , the stress singularity is less severe than one half and, again, this will tend to retard the growth at the vertex until the crack front straightens out.

Macsyma was extensively used throughout the analysis, especially in the generation and use of special functions such as Legendre functions, Gamma functions, Bessel functions from Share directory, in the summation of series, in the solution of linear equations, in seeking roots of determinants of matrices, in the plot routine and in the creation of file for Batch with Teco, etc. This investigation would have been extremely tedious without Macsyma. The methods as well as results in the full entirety, to our knowledge, do not seem to have appeared in literature.

⁹Noble, B., Hussain, M.A., and Pu. S.L., "Apex Singularities for Corner Cracks Under Opening, Sliding, and Tearing Modes," to be published in the Proceedings of International Conf. on Fracture Mechanics in Engr Application, Bangalore, India, 1979.

REFERENCES

1. Barlett, C. C. and Noble, B., "A Variational Method for the Solution of Eigenvalue Problems Involving Mixed Boundary Conditions," Applied Science Research, Section B. Vol. 9, 1962.
2. Green, A. E. and Zerna, W., THEORETICAL ELASTICITY, 2nd Edition, Oxford, 1968.
3. Lure', A. T., THREE DIMENSIONAL PROBLEMS OF THE THEORY OF ELASTICITY, Interscience Publishers, 1964.
4. Noble, B., "The Potential and Charge Distribution Near the Tip of a Flat Angular Sector," EM-135, New York University, NY, 1959.
5. Brown, S. N., and Stewartson, K., "Flow Near the Apex of a Plane Delta Wing," Journal of Institute of Mathematics and Its Applications, Vol. 5, p. 206, 1969.
6. Morrison, J. A. and Lewis, J. A., "Charge Singularity at t7e Corner of a Flat Plate," SIAM, Journal of Applied Mathematics, Vol. 31, p. 233, 1976.
7. Magnus, W. and Oberhettinger, F., SPECIAL FUNCTIONS OF MATHEMATICAL PHYSICS, Chelsea Publishing Co., 1949.
8. Jolly, L. B. W., SUMMATION OF SERIES, Dover Publications, p. 126, 1961.
9. Noble, B., Hussain, M. A., and Pu, S. L., "Apex Singularities for Corner Cracks Under Opening, Sliding and Tearing Modes," to be published in the Proceedings of International Conference on Fracture Mechanics in Engineering Application, Bangalore, India, 1979.

APPENDIX 1

In this appendix we give components of displacement and stress in terms of Papkovitch stress functions ϕ , ψ , ω , and λ . A subscript to a stress function means the partial derivative of the stress function with respect to the variable represented by the subscript, e.g., $\phi_r = \partial\phi/\partial r$, $\phi_{rp} = \partial(\partial\phi/\partial r)/\partial p$. The variable $p = \cos \theta$ is used in the place of θ in both Appendix 1 and Appendix 2. The notation $\bar{p} = \sin \theta = (1-p^2)^{1/2}$ is also used.

The First Basic Solution:

$$2Gu_r = \phi_r$$

$$2Gu_\theta = -\bar{p}\phi_p/r$$

$$2Gu_\phi = \phi_\phi/(r\bar{p})$$

$$\sigma_r = \phi_{rr}$$

$$\sigma_\theta = r^{-2}(\bar{p}^2\phi_{pp} - p\phi_p + r\phi_r)$$

$$\sigma_\phi = r^{-2}(\phi_{\phi\phi}/\bar{p}^2 + r\phi_r - p\phi_p)$$

$$\tau_{\theta\phi} = -r^{-2}(\phi_{p\phi} + p\phi_\phi/\bar{p}^2)$$

$$\tau_{r\phi} = r^{-2}\bar{p}^{-1}(r\phi_{r\phi} - \phi_\phi)$$

$$\tau_{r\theta} = r^{-2}\bar{p}(-r\phi_{rp} + \phi_p)$$

The Second Basic Solution:

$$2Gu_r = [r\psi_r - (3-4\nu)\psi]\bar{p} \cos \theta$$

$$2Gu_\theta = -[\bar{p}^2\psi_p + (3-4\nu)p\psi]\cos \phi$$

$$2Gu_\phi = \psi_\phi \cos \phi + (3-4\nu)\psi \sin \phi$$

$$\sigma_r = [r\psi_{rr} - 2(1-\nu)\psi_r + 2\nu r^{-1}p\psi_p]\bar{p} \cos \phi + 2\nu(rp)^{-1}\psi_\phi \sin \phi$$

$$\begin{aligned}
\sigma_\theta &= r^{-1} [\bar{p}^2 \psi_{pp} + (1-2\nu) p \psi_p + (1-2\nu) r \psi_r] \bar{p} \cos \phi + 2\nu (r\bar{p})^{-1} \psi_\phi \sin \phi \\
\sigma_\phi &= (r\bar{p})^{-1} [\psi_{\phi\phi} + (1-2\nu) r \bar{p}^2 \psi_r - (1-2\nu) p \bar{p}^2 \psi_p] \cos \phi + 2(1-\nu) (r\bar{p})^{-1} \psi_\phi \sin \phi \\
\tau_{\theta\phi} &= (-r\bar{p})^{-1} [\bar{p}^2 \psi_{p\phi} + 2(1-\nu) p \psi_\phi] \cos \phi - (1-2\nu) r^{-1} \bar{p} \psi_p \sin \phi \\
\tau_{r\phi} &= r^{-1} [r \psi_{\phi r} - 2(1-\nu) \psi_\phi] \cos \phi + (1-2\nu) \psi_r \sin \phi \\
\tau_{r\theta} &= [-\bar{p}^2 \psi_{rp} - (1-2\nu) p \psi_r + 2(1-\nu) r^{-1} \bar{p}^2 \psi_p] \cos \phi
\end{aligned}$$

The Third Basic Solution:

$$\begin{aligned}
2Gu_r &= [r\omega_r - (3-4\nu)\omega] \bar{p} \sin \phi \\
2Gu_\theta &= - [\bar{p}^2 \omega_p + (3-4\nu) p \omega] \sin \phi \\
2Gu_\phi &= \omega_\phi \sin \phi - (3-4\nu) \omega \cos \phi \\
\sigma_r &= [r\omega_{rr} - 2(1-\nu)\omega_r + 2\nu r^{-1} p \omega_p] \bar{p} \sin \phi - 2\nu (r\bar{p})^{-1} \omega_\phi \cos \phi \\
\sigma_\theta &= r^{-1} [\bar{p}^2 \omega_{pp} + (1-2\nu) p \omega_p + (1-2\nu) r \omega_r] \bar{p} \sin \phi - 2\nu (r\bar{p})^{-1} \omega_\phi \cos \phi \\
\sigma_\phi &= r^{-1} [\omega_{\phi\phi} / \bar{p}^2 + (1-2\nu) r \omega_r - (1-2\nu) p \omega_p] \bar{p} \sin \phi - 2(1-\nu) (r\bar{p})^{-1} \omega_\phi \cos \phi \\
\tau_{\theta\phi} &= (-r\bar{p})^{-1} [\bar{p}^2 \omega_{p\phi} + 2(1-\nu) p \omega_\phi] \sin \phi + (1-2\nu) r^{-1} \bar{p} \omega_p \cos \phi \\
\tau_{r\phi} &= r^{-1} [r \omega_{r\phi} - 2(1-\nu) \omega_\phi] \sin \phi - (1-2\nu) \omega_r \cos \phi \\
\tau_{r\theta} &= [-\bar{p}^2 \omega_{rp} - (1-2\nu) p \omega_r + 2(1-\nu) r^{-1} \bar{p}^2 \omega_p] \sin \phi
\end{aligned}$$

The Fourth Basic Solution:

$$\begin{aligned}
2Gu_r &= [r\lambda_r - (3-4\nu)\lambda] p \\
2Gu_\theta &= [-p\lambda_p + (3-4\nu)\lambda] \bar{p} \\
2Gu_\phi &= p\bar{p}^{-1} \lambda_\phi \\
\sigma_r &= r p \lambda_{rr} - 2(1-\nu) p \lambda_r - 2\nu r^{-1} \bar{p}^2 \lambda_p \\
\sigma_\theta &= r^{-1} p \bar{p}^2 \lambda_{pp} + (1-2\nu) p \lambda_r + r^{-1} (p^2 + 2\nu \bar{p}^2 - 2) \lambda_p \\
\sigma_\phi &= r^{-1} [p \bar{p}^{-2} \lambda_{\phi\phi} + (1-2\nu) r p \lambda_r - (p^2 + 2\nu \bar{p}^2) \lambda_p]
\end{aligned}$$

$$\tau_{\theta\phi} = r^{-1}[-p\lambda_{p\phi} + (2-2v-\bar{p}^{-2})\lambda_{\phi}]$$

$$\tau_{r\phi} = (r\bar{p})^{-1}[rp\lambda_{\phi r} - 2(1-v)p\lambda_{\phi}]$$

$$\tau_{r\theta} = \bar{p}[-p\lambda_{rp} + 2(1-v)r^{-1}p\lambda_p + (1-2v)\lambda_r]$$

APPENDIX 2

In the following solutions, the selection of $\cos m\phi$ or $\sin m\phi$ depends on the geometry of the problem. The sign on top goes with the trigonometric function on the top and vice versa.

Solution [A]:

$$2Gu_r = (\mu+1)r^\mu p_{\mu+1}^m \begin{matrix} \cos m\phi \\ \sin m\phi \end{matrix}$$

$$2Gu_\theta = (r^\mu/\bar{p}) [(\mu+1)p_{\mu+1}^m - (m+\mu+1)p_\mu^m] \begin{matrix} \cos m\phi \\ \sin m\phi \end{matrix}$$

$$2Gu_\phi = \mp (r^\mu/\bar{p}) m p_{\mu+1}^m \begin{matrix} \sin m\phi \\ \cos m\phi \end{matrix}$$

$$\sigma_r = \mu(\mu+1)r^{\mu-1}p_{\mu+1}^m \begin{matrix} \cos m\phi \\ \sin m\phi \end{matrix}$$

$$\sigma_\theta = (r^{\mu-1}/\bar{p}^2) \{m^2 - \mu - 1 - \bar{p}^2 \mu(\mu+1)\} p_{\mu+1}^m + p(m+\mu+1)p_\mu^m \begin{matrix} \cos m\phi \\ \sin m\phi \end{matrix}$$

$$\sigma_\phi = (r^{\mu-1}/\bar{p}^2) [(-m^2 + \mu + 1)p_{\mu+1}^m - p(m+\mu+1)p_\mu^m] \begin{matrix} \cos m\phi \\ \sin m\phi \end{matrix}$$

$$\tau_{\theta\phi} = \mp (r^{\mu-1}/\bar{p}^2) m [\mu p_{\mu+1}^m - (m+\mu+1)p_\mu^m] \begin{matrix} \sin m\phi \\ \cos m\phi \end{matrix}$$

$$\tau_{r\phi} = \mp (r^{\mu-1}/\bar{p}) m \mu p_{\mu+1}^m \begin{matrix} \sin m\phi \\ \cos m\phi \end{matrix}$$

$$\tau_{r\theta} = (r^{\mu-1}/\bar{p}) \mu [(\mu+1)p_{\mu+1}^m - (m+\mu+1)p_\mu^m] \begin{matrix} \cos m\phi \\ \sin m\phi \end{matrix}$$

Solution [B]:

$$2Gu_r = r^\mu (\mu - 3 + 4\nu) [(-m + \mu + 1)p_{\mu+1}^m - (m + \mu + 1)p_\mu^m] \begin{matrix} \cos m\phi \\ \sin m\phi \end{matrix}$$

$$2Gu_\theta = (r^\mu/\bar{p}) \{(-m - 4 + 4\nu)(-m + \mu + 1)p_{\mu+1}^m + (m + \mu + 1)[m + 4 - 4\nu - \bar{p}^2(\mu + 4 - 4\nu)]p_\mu^m\} \begin{matrix} \cos m\phi \\ \sin m\phi \end{matrix}$$

$$\begin{aligned}
2Gu_{\phi} &= \mp (r^{\mu}/\bar{p}) (m+4-4\nu) [(-m+\mu+1)P_{\mu+1}^m - (m+\mu+1)pP_{\mu}^m] \frac{\sin m\phi}{\cos m\phi} \\
\sigma_r &= r^{\mu-1} [(-m+\mu+1)(\mu^2-3\mu+2\nu-2m\nu-2\nu)P_{\mu+1}^m + (m+\mu+1)(-\mu^2+3\mu+2\nu)pP_{\mu}^m] \frac{\cos m\phi}{\sin m\phi} \\
\sigma_{\theta} &= (r^{\mu-1}/\bar{p}^2) \{ (-m+\mu+1) [(-\mu^2-2\mu\nu-(3-2\nu)(m+1)\bar{p}^2 + (m^2+(5-4\nu)m+4-4\nu)]P_{\mu+1}^m \\
&\quad + (m+\mu+1)p[-m^2-5m+4m\nu-4+4\nu+\bar{p}^2(\mu^2+3\mu+3-2\nu)]P_{\mu}^m \} \frac{\cos m\phi}{\sin m\phi} \\
\sigma_{\phi} &= (r^{\mu-1}/\bar{p}^2) \{ (-m+\mu+1) [-m^2-5m+4m\nu-4+4\nu+\bar{p}^2(1-2\nu)(m+\mu+1)]P_{\mu+1}^m \\
&\quad + (m+\mu+1)p[m^2+5m-4m\nu+4-4\nu-(1-2\nu)(2\mu+1)\bar{p}^2]P_{\mu}^m \} \frac{\cos m\phi}{\sin m\phi} \\
\tau_{\theta\phi} &= \pm (r^{\mu-1}/\bar{p}^2) \{ (-m+\mu+1)p[m^2+(5-4\nu)m+4(1-\nu)]P_{\mu+1}^m \\
&\quad + (m+\mu+1)[\bar{p}^2(m\mu+(3-2\nu)m+2(1-\nu)\mu+4(1-\nu))-m^2-(5-4\nu)m-4(1-\nu)]P_{\mu}^m \} \frac{\sin m\phi}{\cos m\phi} \\
\tau_{r\phi} &= \mp (r^{\mu-1}/\bar{p}) [m(\mu-2+2\nu)+2\mu(1-\nu)-2(1-\nu)] [(-m+\mu+1)P_{\mu+1}^m - (m+\mu+1)pP_{\mu}^m] \frac{\sin m\phi}{\cos m\phi} \\
\tau_{r\theta} &= (r^{\mu-1}/\bar{p}) \{ (-m+\mu+1)p[-m\mu+2(1-\nu)(m-\mu+1)]P_{\mu+1}^m \\
&\quad + (m+\mu+1)[m\mu+2(1-\nu)(-m+\mu-1)+\bar{p}^2(-\mu^2+2-2\nu)]P_{\mu}^m \} \frac{\cos m\phi}{\sin m\phi}
\end{aligned}$$

Solution [C]:

$$\begin{aligned}
2Gu_r &= r^{\mu}(\mu-3+4\nu)(-P_{\mu+1}^m + pP_{\mu}^m) \frac{\cos m\phi}{\sin m\phi} \\
2Gu_{\theta} &= (r^{\mu}/\bar{p}) \{ (-m+4-4\nu)pP_{\mu+1}^m + [(m-4+4\nu)+\bar{p}^2(\mu+4-4\nu)]P_{\mu}^m \} \frac{\cos m\phi}{\sin m\phi} \\
2Gu_{\phi} &= \mp (r^{\mu}/\bar{p}) (m-4+4\nu)(-P_{\mu+1}^m + pP_{\mu}^m) \frac{\sin m\phi}{\cos m\phi} \\
\sigma_r &= r^{\mu-1} \{ [-\mu^2+\mu(3-2\nu)-2m\nu+2\nu]P_{\mu+1}^m + (\mu^2-3\mu-2\nu)pP_{\mu}^m \} \frac{\cos m\phi}{\sin m\phi}
\end{aligned}$$

$$\sigma_{\theta} = (r^{\mu-1}/\bar{p}^2) \{ [\bar{p}^2 (\mu^2 + 2\nu\mu - (3-2\nu)(m-1)) + (-m^2 + (5-4\nu)m - 4(1-\nu))] P_{\mu+1}^m$$

$$+ [\bar{p}^2 (-\mu^2 - 3\mu - 3 + 2\nu) + m^2 - (5-4\nu)m + 4(1-\nu)] P_{\mu}^m \} \begin{matrix} \cos m\phi \\ \sin m\phi \end{matrix}$$

$$\sigma_{\phi} = (r^{\mu-1}/\bar{p}^2) \{ [\bar{p}^2 (1-2\nu)(m-\mu-1) + m^2 - (5-4\nu)m + 4(1-\nu)] P_{\mu+1}^m$$

$$+ [\bar{p}^2 (1-2\nu)(2\mu+1) - m^2 + (5-4\nu)m - 4(1-\nu)] P_{\mu}^m \} \begin{matrix} \cos m\phi \\ \sin m\phi \end{matrix}$$

$$\tau_{\theta\phi} = \pm (r^{\mu-1}/\bar{p}^2) [m^2 - (5-4\nu)m + 4(1-\nu)] P_{\mu+1}^m$$

$$+ [\bar{p}^2 (-m\mu - (3-2\nu)m + 2(1-\nu)\mu + 4(1-\nu)) - m^2 + (5-4\nu)m - 4(1-\nu)] P_{\mu}^m \} \begin{matrix} \sin m\phi \\ \cos m\phi \end{matrix}$$

$$\tau_{r\phi} = \pm (r^{\mu-1}/\bar{p}) [-m\mu + 2(1-\nu)m + 2(1-\nu)(\mu-1)] (-P_{\mu+1}^m + P_{\mu}^m) \begin{matrix} \sin m\phi \\ \cos m\phi \end{matrix}$$

$$\tau_{r\theta} = (r^{\mu-1}/p) \{ [-m\mu + 2(1-\nu)m + 2(1-\nu)(\mu-1)] P_{\mu+1}^m$$

$$+ [m\mu - 2(1-\nu)m - 2(1-\nu)(\mu-1) - p^2 (-\mu^2 + 2 - 2\nu)] P_{\mu}^m \} \begin{matrix} \cos m\phi \\ \sin m\phi \end{matrix}$$

Solution [D]:

$$2Gu_r = (\mu - 3 + 4\nu) r^{\mu} p P_{\mu}^m \begin{matrix} \cos m\phi \\ \sin m\phi \end{matrix}$$

$$2Gu_{\theta} = (r^{\mu}/\bar{p}) \{ (-m + \mu + 1) P_{\mu+1}^m + [\bar{p}^2 (\mu + 4 - 4\nu) - (\mu + 1)] P_{\mu}^m \} \begin{matrix} \cos m\phi \\ \sin m\phi \end{matrix}$$

$$2Gu_{\phi} = \mp (mp/\bar{p}) r^{\mu} P_{\mu}^m \begin{matrix} \sin m\phi \\ \cos m\phi \end{matrix}$$

$$\sigma_r = r^{\mu-1} \{ 2\nu(-m + \mu + 1) P_{\mu+1}^m + (\mu^2 - 3\mu - 2\nu) P_{\mu}^m \} \begin{matrix} \cos m\phi \\ \sin m\phi \end{matrix}$$

$$\sigma_{\theta} = (r^{\mu-1}/\bar{p}^2) \{ (m - \mu - 1) [1 - (3-2\nu)\bar{p}^2] P_{\mu+1}^m + [m^2 + \mu + 1 + \bar{p}^2 (-\mu^2 - 3\mu - 3 + 2\nu)] P_{\mu}^m \} \begin{matrix} \cos m\phi \\ \sin m\phi \end{matrix}$$

$$\sigma_{\phi} = (r^{\mu-1}/\bar{p}^2) \{ (-m + \mu + 1) [1 + \bar{p}^2 (-1 + 2\nu)] P_{\mu+1}^m + [-m^2 - \mu - 1 + (1-2\nu)(2\mu+1)\bar{p}^2] P_{\mu}^m \} \begin{matrix} \cos m\phi \\ \sin m\phi \end{matrix}$$

$$\tau_{\theta\phi} = \pm (r^{\mu-1}/\bar{p}^2)m\{(m-\mu-1)pP_{\mu+1}^m + [\mu+2-\bar{p}^2(\mu+3-2\nu)]P_{\mu}^m\} \begin{matrix} \sin m\phi \\ \cos m\phi \end{matrix}$$

$$\tau_{r\phi} = \pm (2-2\nu-\mu)mp(r^{\mu-1}/\bar{p})P_{\mu}^m \begin{matrix} \sin m\phi \\ \cos m\phi \end{matrix}$$

$$\tau_{r\theta} = (r^{\mu-1}/\bar{p})\{(-m+\mu+1)(\mu-2+2\nu)pP_{\mu+1}^m + [-\mu^2+\mu(1-2\nu)+2(1-\nu) + \bar{p}^2(\mu^2-2+2\nu)]P_{\mu}^m\} \begin{matrix} \cos m\phi \\ \sin m\phi \end{matrix}$$

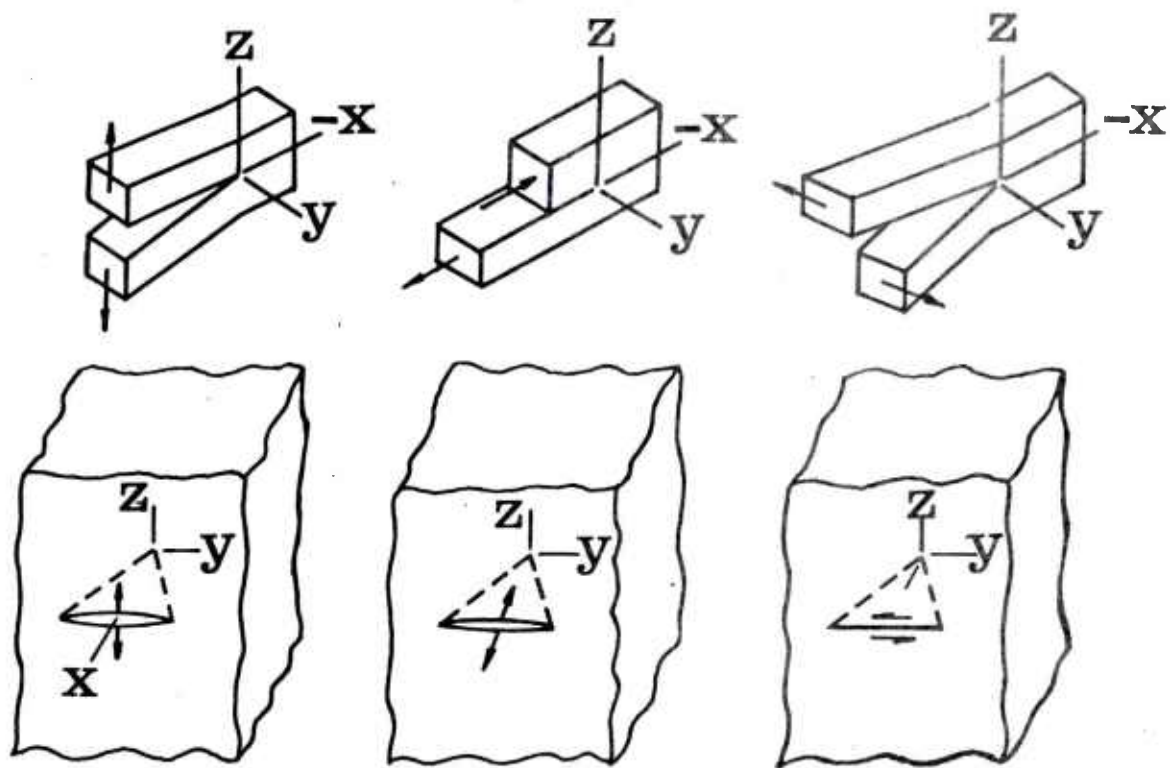


Figure 1. A flat wedge-shaped crack under three different modes and its two-dimensional counterparts.

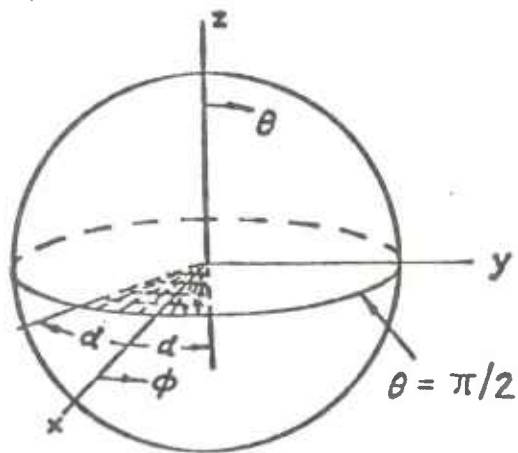


Figure 2. A spherical coordinate surrounding the apex of a thin angular sector crack.

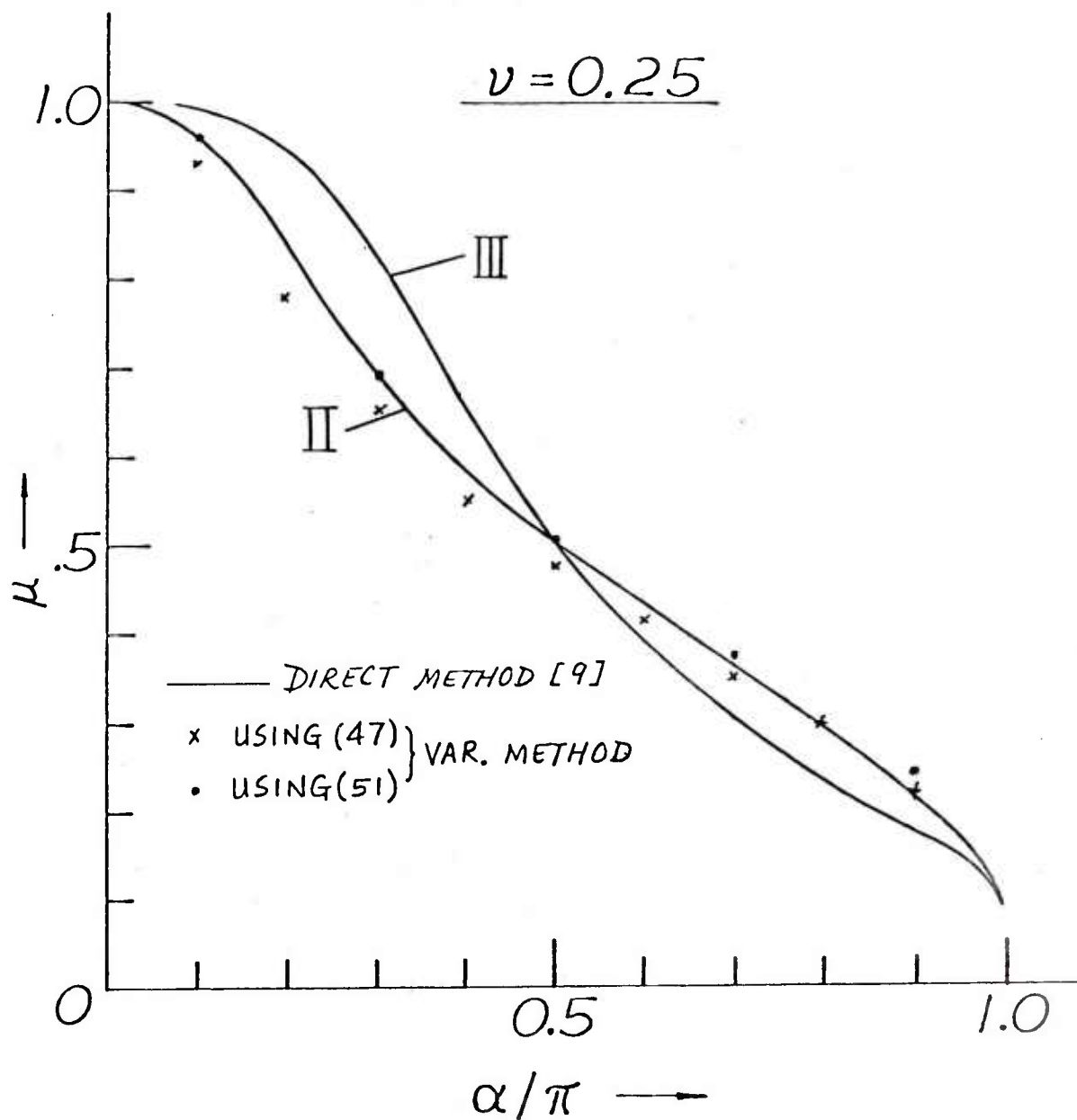


Figure 3. The eigenvalue μ as a function of α/π for $\nu = 0.25$.

(a) Variational results indicated by x and • for mode II.

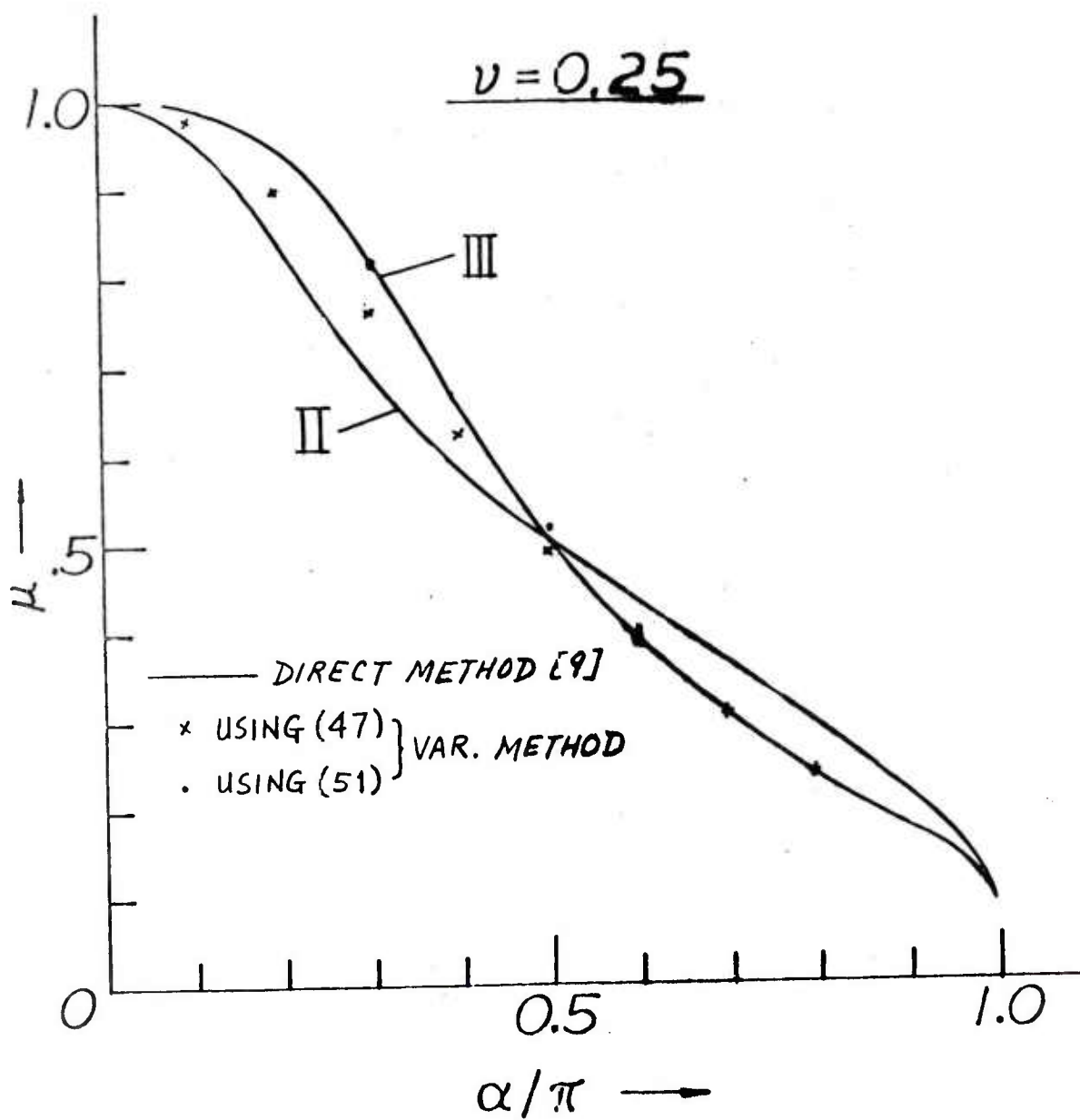


Figure 3(b). Variational results indicated by x and • for mode III.

TECHNICAL REPORT INTERNAL DISTRIBUTION LIST

	<u>NO. OF COPIES</u>
COMMANDER	1
CHIEF, DEVELOPMENT ENGINEERING BRANCH	1
ATTN: DRDAR-ICB-DA	1
-DM	1
-DP	1
-DR	1
-DS	1
-DC	1
CHIEF, ENGINEERING SUPPORT BRANCH	1
ATTN: DRDAR-ICB-SE	1
-SA	1
CHIEF, RESEARCH BRANCH	2
ATTN: DRDAR-ICB-RA	1
-RC	1
-RM	1
-RP	1
CHIEF, LWC MORTAR SYS. OFC.	1
ATTN: DRDAR-ICM	1
CHIEF, IMP. 81MM MORTAR OFC.	1
ATTN: DRDAR-ICB-I	1
TECHNICAL LIBRARY	5
ATTN: DRDAR-ICB-TL	
TECHNICAL PUBLICATIONS & EDITING UNIT	2
ATTN: DRDAR-ICB-TL	
DIRECTOR, OPERATIONS DIRECTORATE	1
DIRECTOR, PROCUREMENT DIRECTORATE	1
DIRECTOR, PRODUCE ASSURANCE DIRECTORATE	1

NOTE: PLEASE NOTIFY ASSOC. DIRECTOR, BENET WEAPONS LABORATORY, ATTN:
DRDAR-ICB-TL, OF ANY REQUIRED CHANGES.

TECHNICAL REPORT EXTERNAL DISTRIBUTION LIST

	<u>NO. OF COPIES</u>		<u>NO. OF COPIES</u>
ASST SEC OF THE ARMY RESEARCH & DEVELOPMENT ATTN: DEP FOR SCI & TECH THE PENTAGON WASHINGTON, D.C. 20315	1	COMMANDER US ARMY TANK-AUTMV R&D CMD ATTN: TECH LIB - DRDTA-UL MAT LAB - DRDTA-RK WARREN MICHIGAN 48090	1 1
COMMANDER US ARMY MAT DEV & READ. CMD ATTN: DRCDE 5001 EISENHOWER AVE ALEXANDRIA, VA 22333	1	COMMANDER US MILITARY ACADEMY ATTN: CHMN, MECH ENGR DEPT WEST POINT, NY 10996	1
COMMANDER US ARMY ARRADCOM ATTN: DRDAR-IC -ICA (PLASTICS TECH EVAL CEN) -ICE -LCM -ICS -LCW -TSS(STINFO) DOVER, NJ 07801	1 1 1 1 1 2	COMMANDER REDSTONE ARSENAL ATTN: DRSMI-RB -RRS -RSM ALABAMA 35809 COMMANDER ROCK ISLAND ARSENAL ATTN: SARRI-ENM (MAT SCI DIV) ROCK ISLAND, IL 61202	2 1 1 1 1
COMMANDER US ARMY ARRCOM ATTN: DRSAR-LEP-L ROCK ISLAND ARSENAL ROCK ISLAND, IL 61299	1	COMMANDER HQ, US ARMY AVN SCH ATTN: OFC OF THE LIBRARIAN FT RUCKER, ALABAMA 36362	1
DIRECTOR US Army Ballistic Research Laboratory ATTN: DRDAR-TSB-S (STINFO) ABERDEEN PROVING GROUND, MD 21005.	1	COMMANDER US ARMY FGN SCIENCE & TECH CEN ATTN: DRXST-SD 220 7TH STREET, N.E. CHARLOTTESVILLE, VA 22901	1
COMMANDER US ARMY ELECTRONICS CMD ATTN: TECH LIB FT MONMOUTH, NJ 07703	1	COMMANDER US ARMY MATERIALS & MECHANICS RESEARCH CENTER ATTN: TECH LIB - DRXMR-PL WATERTOWN, MASS 02172	2
COMMANDER US ARMY MOBILITY EQUIP R&D CMD ATTN: TECH LIB FT BELVOIR, VA 22060			

NOTE: PLEASE NOTIFY COMMANDER, ARRADCOM, ATTN: BENET WEAPONS LABORATORY, DRDAR-ICB-TL, WATERVLIET ARSENAL, WATERVLIET, N.Y. 12189, OF ANY REQUIRED CHANGES.

TECHNICAL REPORT EXTERNAL DISTRIBUTION LIST (CONT)

	NO. OF COPIES		NO. OF COPIES
COMMANDER US ARMY RESEARCH OFFICE P.O. BOX 12211 RESEARCH TRIANGLE PARK, NC 27709	1	COMMANDER DEFENSE TECHNICAL INFO CENTER ATTN: DTIA-TCA CAMERON STATION ALEXANDRIA, VA 22314	12
COMMANDER US ARMY HARRY DIAMOND LAB ATTN: TECH LIB 2800 POWDER MILL ROAD ADELPHIA, MD 20783	1	METALS & CERAMICS INFO CEN BATTELLE COLUMBUS LAB 505 KING AVE COLUMBUS, OHIO 43201	1
DIRECTOR US ARMY INDUSTRIAL BASE ENG ACT ATTN: DRXPE-MT ROCK ISLAND, IL 61201	1	MECHANICAL PROPERTIES DATA CTR BATTELLE COLUMBUS LAB 505 KING AVE COLUMBUS, OHIO 43201	1
CHIEF, MATERIALS BRANCH US ARMY R&S GROUP, EUR BOX 65, FPO N.Y. 09510	1	MATERIEL SYSTEMS ANALYSIS ACTV ATTN: DRXSY-MP ABERDEEN PROVING GROUND MARYLAND 21005	1
COMMANDER NAVAL SURFACE WEAPONS CEN ATTN: CHIEF, MAT SCIENCE DIV DAHLGREN, VA 22448	1		
DIRECTOR US NAVAL RESEARCH LAB ATTN: DIR, MECH DIV CODE 26-27 (DOC LIB) WASHINGTON, D. C. 20375	1 1		
NASA SCIENTIFIC & TECH INFO FAC P. O. BOX 8757, ATTN: ACQ BR BALTIMORE/WASHINGTON INTL AIRPORT MARYLAND 21240	1		

NOTE: PLEASE NOTIFY COMMANDER, ARRADCOM, ATTN: BENET WEAPONS LABORATORY,
DRDAR-LCB-TL, WATERVLIET ARSENAL, WATERVLIET, N.Y. 12189, OF ANY
REQUIRED CHANGES.